



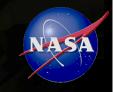


### **Towed Glider Air Launch System**

Prepared by: Jerry Budd, PM Angie Monheim, PM Support NASA Armstrong Flight Research Center Edwards, CA

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### GOAL



### Develop an affordable system that truly enables



- So far, Launch-on-Demand has not come to fruition because three key elements are not yet responsive in and of themselves
  - "On the shelf" satellites ready to be deployed
  - "On the shelf" launch vehicles ready to be deployed
  - Flexible launch platform/range with quick turn-around times
- These elements are getting very close to being realizable, but they
  need a little help to get momentum going in the right direction

An enabling function is needed to reach the tipping point...

### **APPROACH**



Three elements must be combined to provide the <u>enabling function</u>

### **STAGE 1: PLATFORM**

Demonstrate a remotely piloted **GLIDER** carrying a small LV **TOWED** by a minimally modified business jet

### **STAGE 2: LV STIMULUS**

Foster integration of several small LV companies with program to demonstrate launch off the **TOWED GLIDER** 

### **BEHIND THE SCENES: POLICY**

Work with FAA to make launch policies for small LVs using **TOWED GLIDER** as the launch platform

LV = Launch Vehicle







Ignite the small satellite launch market and make Operationally Responsive Space (ORS) and Launch On Demand a reality

### STAGE 1: PLATFORM



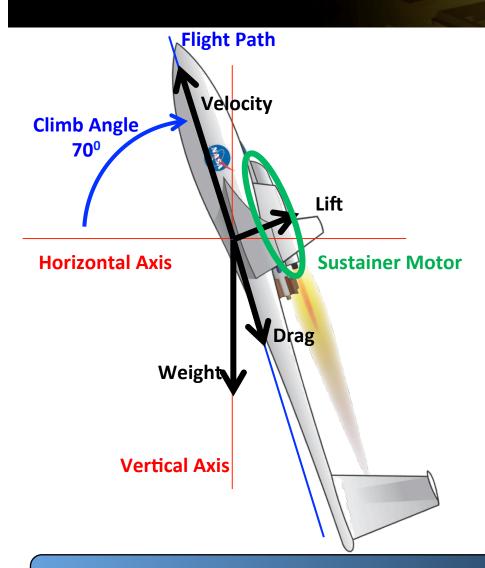
3 Year project to build a 100' wingspan remotely-piloted glider, modify an existing business jet for tow, and release 6000 lb LV with 100 lb satellite at 40K', 70°, M=0.8 safely & effectively.



Simply put, the Towed Glider concept is the **OPTIMIZATION** of air launch

### **Sustainer Motor**





**Location:** Mounted on top of the glider

**Purpose:** Provides variable thrust on demand to change the Glider and LV orientation from horizontal to nearly vertical

#### **Features:**

- Restartable
- Throttleable down to ~15% or less
- Controllable
- Increases Glider fly-back range

**Profile:** Start horizontal, idle sustainer motor, begin pull-up towards 70° climb, use sustainer motor variable thrust to maintain constant airspeed during climb, stabilize at 70° then release LV

The sustainer motor provides the energy to go from horizontal to nearly vertical so the LV is optimally oriented for launch

Flight demonstrating the sustainer motor concept at 1/3 scale would increase the TRL from 3 to 5 while simultaneously reducing programmatic and technical risk for the full scale program

# PLATFORM: Why Towed Glider?



- High Performance: 50% plus increase in performance to orbit
  - Carry 2+ times mass to altitude as same size direct carry conventional aircraft
  - Pull-up maneuver gives 12-15% performance increase over horizontal release
- Low Cost
  - No new infrastructure costs.
  - Relatively inexpensive to build a glider versus a conventional airplane
  - Low maintenance costs compared to a conventional airplane
  - Many existing business jets are candidate tow vehicles with minimal mods
  - Easier for small LV companies to get started without huge capital outlays
  - Commercialize-able as a separate service from LVs
- Safety & Mission Assurance
  - Unmanned glider eliminates human concerns for carrying LV
  - Restartable sustainer motor extends glide home distance following launch or abort
  - Glider capable of landing with LV attached in event of mission abort
- Flexibility
  - Glider plug-n-play center wing allows multiple simultaneous LV build-ups
  - Glider and LV could fit inside C-17 tow plane for transport to desired airfield
  - Inexpensive gliders can be staged at any airfield, ready for immediate launch
  - Tow plane can be existing aircraft that simply adds towed launch to duties
  - Glider concept is scalable from very small LVs

The Towed Glider concept is the **OPTIMIZATION** of air launch

# Carry Efficiency for Existing Air-Launch Platforms



Launch Platform	Gross Takeoff Weight (GTOW)	Carried Weight	Carry Efficiency, $\frac{W_{carried}}{(GTOW - W_{carried})}$
WhiteKnightTwo <sup>(2)</sup>	70×10³ lb	29×10³ lb	0.71
Stratolaunch <sup>(3)</sup>	1.3×10 <sup>6</sup> lb	500×10³ lb	0.63
B-52 Stratofortress <sup>(4)</sup>	488×10³ lb	70×10³ lb	0.17
L-1011 Stargazer	430×10³ lb <sup>(5)</sup>	52×10 <sup>3</sup> lb <sup>(6)</sup>	0.14

Key Performance Parameter				
Performance Parameter	State of the Art	Threshold Value	Project Goal	
Carry Efficiency	0.71	1.5	2.0	

<sup>&</sup>lt;sup>2</sup> From FAA Draft Environmental Assessment for the Launch and Re-entry of SpaceShipTwo; <a href="http://www.faa.gov/about/office-org/headquarters-offices/ast/media/20120309-Mojave-SS2-Draft-EA.pdf">http://www.faa.gov/about/office-org/headquarters-offices/ast/media/20120309-Mojave-SS2-Draft-EA.pdf</a>.

Carried weight represents SS2 maximum launch weight; the maximum carry weight of the WK2 could not be found in the public record.

<sup>&</sup>lt;sup>3</sup> Stratolaunch Press Kit; http://www.stratolaunch.com/presskit/Stratolaunch\_PressKitFull\_May2013.pdf.

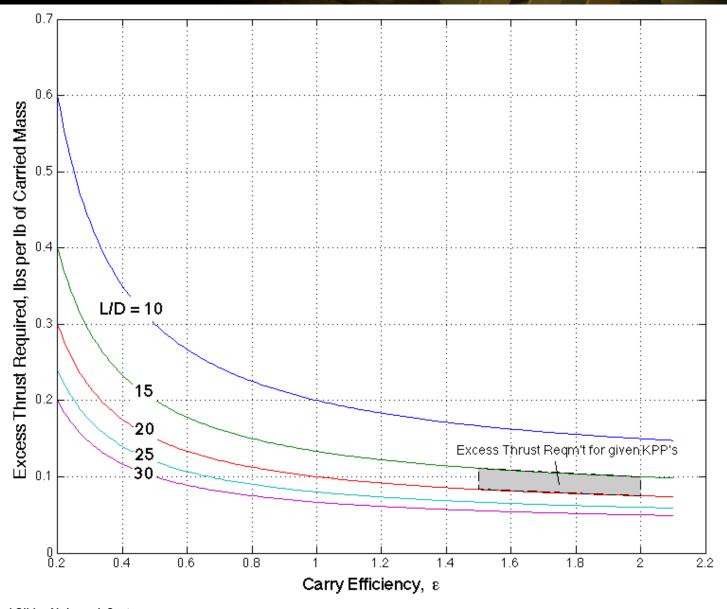
<sup>&</sup>lt;sup>4</sup> Air Force Fact Sheet; http://www.af.mil/AboutUs/FactSheets/Display/tabid/224/Article/104465/b-52- stratofortress.aspx.

<sup>&</sup>lt;sup>5</sup> The most favorable (lightest) GTOW of the L-1011 variants; taken from http://en.wikipedia.org/wiki/ Lockheed\_L-1011\_TriStar

<sup>&</sup>lt;sup>6</sup> Orbital L-1011 Fact Sheet; http://www.orbital.com/LaunchSystems/Publications/L1011\_factsheet.pdf

# Excess Thrust Required by Tow Airplane as a function of Carry Efficiency and Tow Glider Max L/D





# White Knight II: Direct Carry vs Towing Glider



### Real-world example of why Towing is a better business case

- Question: How much more could White Knight II (WK2) bring to altitude if Towing a Glider?
  - WK2 direct lift capability = 30,000 lbs
  - WK2 excess thrust ~ 4900 lbs (assumes 100% tow efficiency)
  - Rocket to glider weight ratio = 2:1 (rocket twice as heavy as glider)
  - Glider (with rocket attached) with 20:1 Lift to Drag ratio (L/D)
- Math: WK2 towing 20:1 L/D glider and rocket 2x weight of glider
  - Excess Thrust X L/D X 2/3 = 4900 X 20 X 2/3 = 65,333 lbs
  - 65,333 lbs / 30,000 lbs = 2.18 times direct carry capability

Regardless of system size, using a towed glider allows us to launch 2X as much to orbit for less money than a comparably sized direct carry aircraft

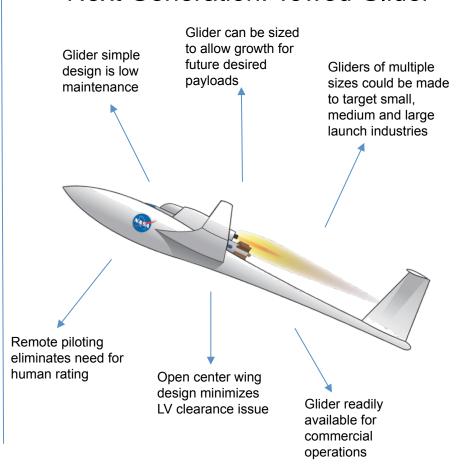
# Glider Design Creates Trade Space



### Current State of the Art: Direct Carry

### Moderate to high maintenance Hard weight limit for requirements LV means no room Payload max size for growth/error dictated by limits of base aircraft More extensive separation analysis required Limited space High performance jets under Aircraft not readily available fuselage for commercial use

### **Next Generation: Towed Glider**

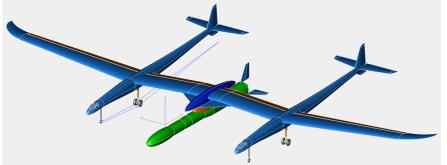


Glider adds flexibility to ensure design success and commercializability

# PLATFORM: SAS-ASI-Burt Rutan Study



- NASA contracted Special Aerospace Services (SAS) to study the viability of the Towed Glider Air Launch Concept
  - SAS teamed with Aerospace Services
     International and Rutan Designs
  - 120 Day Feasibility Study Conducted in 2012
  - Results presented in January 2013
    - Burt Rutan assessed the concept feasibility and designed a conceptual air-launch towed glider
    - Ken Lindas (SAS) analyzed the improvements in launch vehicle payload performance to orbit



Design Carry Efficiency: 1.5

Excerpts from the SAS Study Presentation:

"This world class design team was surprised in the performance enhancement relative to a Towed Carrier Aircraft leveraging an existing Launch Vehicle System."

"Viability should be considered for demonstration..."

"This is potentially game changing research with multiple Industry and USG applications."

The study showed the concept is do-able...next step is the Proof of Concept

### 1/3 Scale Model Glider



 FY13/14 Center Innovation Funds used to develop single fuselage and twin fuselage gliders for tow behind the DROID unmanned model

aircraft

- Remote piloting of glider implemented using the Ground Control Station for DROID
- Whittinghill Aerospace under SBIR contract to build a small sub-orbital LV for launch demo off the twin fuselage glider
- Twin Fuselage Glider towed behind DROID to be used for 1/3 scale sustainer motor risk reduction work in FY15 (requires additional funding)





# STAGE 2: LV Stimulus



**Enables** candidate **small launch vehicle companies** to become **viable commercial launch providers** for putting small satellites into Low Earth Orbit (LEO) **on demand**.

- Stimulate Small Launch Vehicle Market
  - Development companies can enter air launch market without huge capital outlays
  - Successful launch demos could lead to future LSP contracts, as well as contracts with Industry Partners
- Stimulate Small Satellite Development Market
  - Give small satellites an affordable path to orbit without ride sharing
  - Affordable dedicated launch options will encourage new customers for small satellites

The end goal is to produce multiple viable small LV providers for the small launch market...the Glider enables them

### POLICY: Tailored for Small Launch



Collaborative partnership between NASA and FAA to coordinate policies and procedures which address the needs and issues associated with small launch

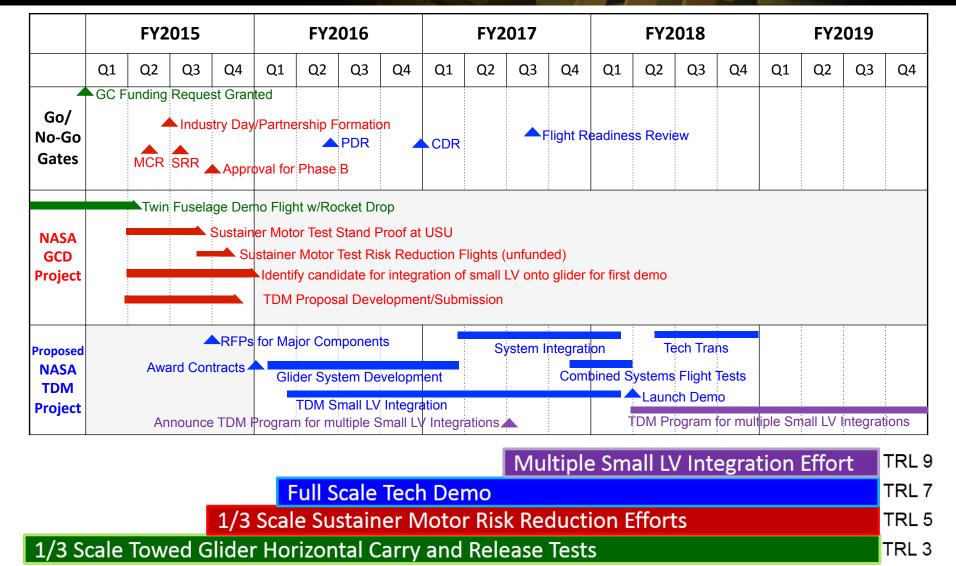
# & REGULATIONS

- Appropriate safety requirements for small LVs
- Certification of tow modifications to existing aircraft
- Policies and Regulations for servicing small LVs at airfields
- Paperwork and processing time that is conducive to ORS/Launch-On-Demand type CONOPS
- Coordinate streamlined conjunction assessment (CA) process and prepare for increased launch assessment traffic
- Orbital Debris Mitigation policies for the expected increase in small satellite population
- International policy concerns

The goal is to ensure NASA, government and military are prepared to accept and support the dedicated small satellite launch industry

# Schedule, Deliverables, and Key Milestones

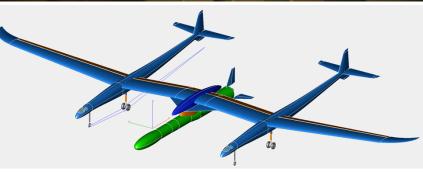




# Summary







Goal: The goal is to enable Operationally Responsive Space. PLATFORM - Remotely piloted GLIDER carrying a small LV TOWED by a minimally modified business jet Approach: LV STIMULUS - Foster integration of several small LV companies with program to launch off TOWED GLIDER POLICY - Work with FAA to make launch policies for small LVs using TOWED GLIDER launch platform NASA GCD Project - NASA provides funding and FTEs Program: Proposed NASA TDM Project - NASA provides FTEs, Partners provide \$\$, LV Candidates provide innovative new small LV technology

> Glider acts as an enabling function to allow small launch vehicles affordable access to optimized air launch conditions they wouldn't normally be able to afford

TRL9 Multiple Small LV Integration Effort **Full Scale Tech Demo** TRL 7 1/3 Scale Sustainer Motor Risk Reduction Efforts TRL 5 1/3 Scale Towed Glider Horizontal Carry and Release Tests TRL 3

2015 2016 2017 2018 2019

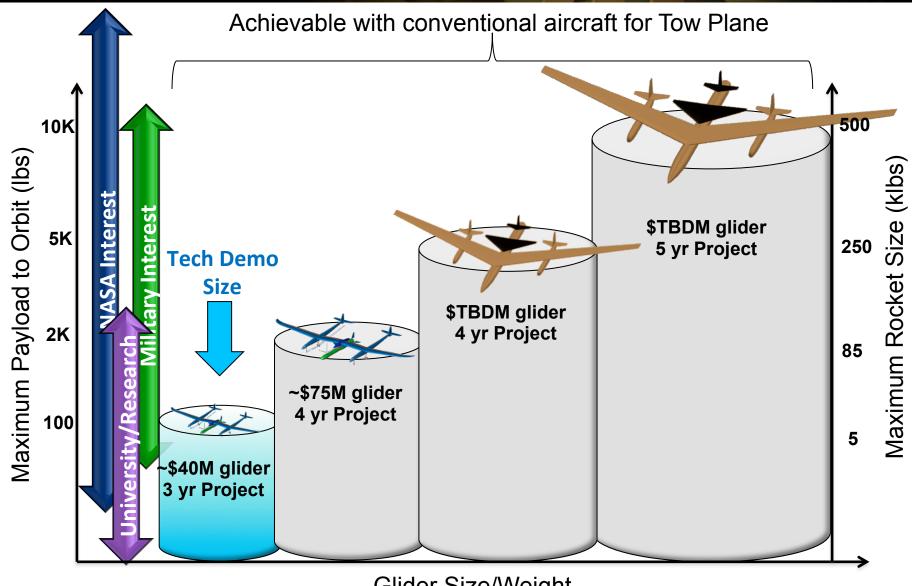
Key:



# **Backup Material**

# Towed Glider Technology is Scalable





Glider Size/Weight

# Launch System Comparison

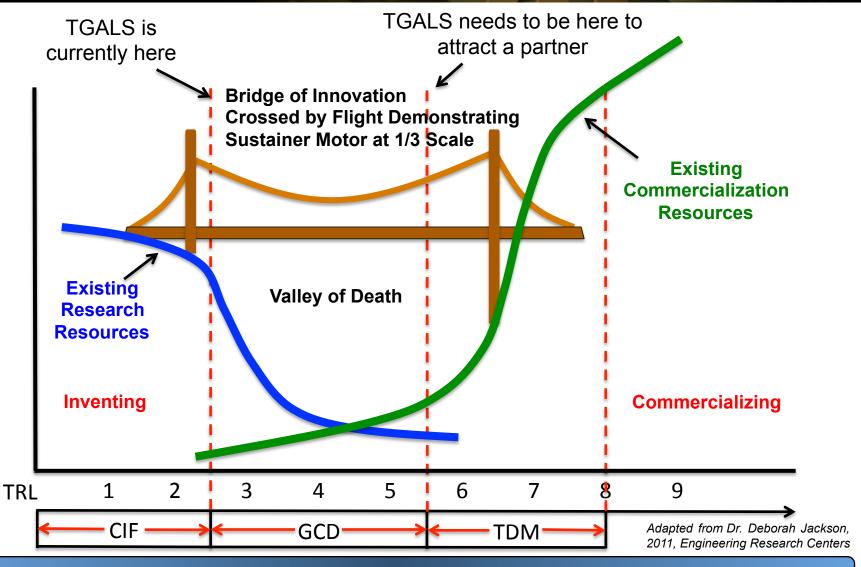


	Large, expensive co				Minimal
Fixed Infrastructure		(A)		G	ST
Maintenance	High				Minimal
Hours/Costs	<b>(3</b> )	A	L G	S	
Scalability (can I	Not Scalable				Highly Scalable
make it bigger?)	AGD		S		(F)
	Months				Minutes
Launch on Demand			<u>(L)</u>	G	AS
Portability/	Limited Locations				Locate virtually anywhere
Survivability	<b>(</b>	A		G	ST
	6000 lbs or less				> Atlas V
LV Size Limit	AG		S	<u> </u>	
Optimized Launch	Fixed Coordinates				Virtually any coordinates
Coordinates	(F)		<u>S</u>	LGA	
LV Performance	Baseline				> 60% Increase
(assumes same LV)	FS			LG	AT
	F	<u>A</u>	(S)	<u> </u>	<u> </u>
TGALS F	Fixed Pad	ALASA	SWORDS	Go-Launcher	Stratolaunch

**Towed Glider Combines Best Attributes of Current Systems** 

# "TRL Jumping" over the Valley of Death





By taking candidate low TRL concepts to flight we "jump" over the Valley of Death in 1-3 yrs

# Game Changing Potential of Towed Glider System



# Enable Operationally Responsive Space (ORS) and Reduce Reliance on Fixed Ranges

- Enable the dedicated small launch market to finally emerge
  - Enables ORS and Launch-on-Demand
    - COCOMs can responsively launch small satellites to meet emergent theater needs
    - Launches can originate from almost any airfield making them hard to predict
  - Enables rapid refresh of on-orbit instruments
    - o Increased viability of space-based internet bringing service to remote areas
    - o Allows optimized Earth Observation of emerging events on Earth
    - o Bring incremental improvements in technology to orbit faster
  - Enables the possibility of cost effective dedicated small launch
- Begin to remove the reliance on fixed ground ranges
  - Help reduce future range fixed infrastructure requirements
  - Help ensure NSS launch survivability if a fixed range is rendered inoperable
  - Help optimize launch conditions for each specific orbit

Potential to Revolutionize the Dynamics of Launch Manifesting in US

# Project Snapshot: Towed Glider Air Launch





### Why a Towed Glider?

- High Performance: 50% plus increase in performance to orbit
  - Carry 2+ times to alt as same size direct carry piloted aircraft
  - Pull-up maneuver gives 12-15% increase over horizontal release
- Low Cost:
  - No new infrastructure costs
  - Many existing candidate tow vehicles
  - Easier for small LV companies to get started w/o huge outlay
  - Commercialize-able as a separate service from LVs
- Safety:
  - Remotely piloted glider eliminates human concerns for LV
  - Glider capable of fly home & landing with LV attached
- Flexibility:
  - Glider plug-n-play center wing allows simultaneous LV build-ups
  - Glider and LV could fit inside C-17 tow plane for transport
  - Inexpensive gliders can be staged at any airfield
  - Glider concept is scalable from very small LVs

#### Goal

Affordable System Enabling Operationally Responsive Space
 Part Program

- PLATFORM Remotely piloted GLIDER carrying a small LV TOWED by a minimally modified business jet
- LV STIMULUS Foster integration of several small LV companies with program to launch off TOWED GLIDER
- POLICY Work with FAA to make launch policies for small LVs using TOWED GLIDER as the launch platform

#### **Notional Partnership**

- GCD NASA provides funding and FTEs
- Tech Demo NASA provides FTEs, Partners provide \$\$, LV Candidates provide innovative new small LV technology

#### TRL Steps

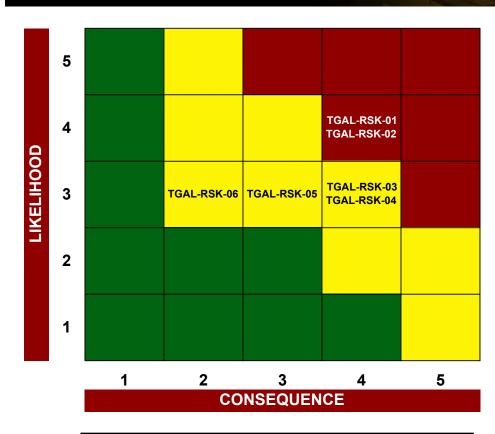
		M	ultiple Small LV In	tegration Effort	TRL9
	Full Scal	e Tech Demo			TRL 7
	1/3 Scale Sustai	ner Motor Risk R	eduction Efforts		TRL 5
1/3 Scale Towed	d Glider Horizontal	Carry and Releas	e Tests		TRL 3
2015	2016	2017	2018	2019	•

#### **Tech Demo Deliverables**

•	Towed Glider System Development & Build-Up Flights	FY16-18
•	Small LV Integration Program	FY16-18
•	Policy Development and Coordination	FY16-18
•	Small Launch Demo off Glider and Tech Transfer	FY18

# Risk Management Matrix





Criticalit	y LxCTrend	<b>Approach</b>
High	▼ Decreasing (Improving)	M - Mitigate
Med	▲ Increasing (Worsening)	W - Watch
Med	Unchanged	A - Accept
Low	New Since Last Period	R - Research

### Risk 1: Performance or operational issues with sustainer motor performance/design

Mitigation 1: Risk-reduction ground tests in FY15 with USU sustainer motor

Mitigation 2: Risk-reduction flight tests in with 1/3 scale glider & USU sustainer motor (currently unfunded)

### Risk 2: Problems with effecting new "Policy" for nontraditional launch range ops

Mitigation: Begin investigation in FY15 of the "Core Level" requirements for performing routine launch to orbit operations outside of traditional DoD/NASA Ranges while fully meeting US and Agency requirements

### Risk 3: Small Launch Vehicle integration delays/ problems

Mitigation: Find and fund mechanism to assist multiple small launch vehicle providers to develop candidate small LV's for air launch from TGALS

### Risk 4: Glider load carrying capability significantly lower than predicted/expected

Mitigation: Risk-reduction aerodynamic performance flight tests in FY14 with 1/3-scale glider

### Risk 5: Glider performance under tow significantly lower than predicted/expected

Mitigation: Risk-reduction dynamic tow characterization flight tests in FY15 with 1/3-scale glider

### Risk 6: Glider aerodynamic performance significantly lower than predicted/expected

Mitigation: Risk-reduction aerodynamic performance flight

# Socialization



Name	Position	Date
Robert Lightfoot Jr. & Lesa Roe	NASA Associate Administrator & Deputy Associate Administrator	11/25/14
Tom King	SORDAC Space – Special Operations Command	11/5/14
Dr. James Reuther & STMD DPMC	Deputy AA for Programs, Space Technology Mission Directorate	9/4/14
Dr. John Grunsfeld	Associate Administrator for the Science Mission Directorate	7/31/14
Michael Lopez-Alegria	President of the Commercial Spaceflight Federation	7/31/14
Dr. George Nield	Associate Administrator for FAA Commercial Space Transportation	7/31/14
Frank DiBello	President and Chief Executive Officer of Space Florida	7/29/14
Jay Dryer and Barbara Esker	Director and Deputy Director of the Fundamental Aeronautics Program, ARMD	6/25/14
Dr. Ellen Stofan	NASA Chief Scientist	6/23/14
Dr. James Reuther	Deputy AA for Programs, Space Technology Mission Directorate	6/20/14
DoD Joint Space Team	Executive forum advising DoD on Space issues	6/19/14
Dr. Alexander MacDonald	Assistant to David Miller, NASA Chief Technologist	6/4/14
Eric Poole	KSC Launch Services Program System Integration Engineer	6/2/14
Charles Bolden & Dr. David Miller	NASA Administrator & NASA Chief Technologist	5/13/14
Stuart Witt	Chief Executive Officer, Mojave Air and Space Port	4/17/14
Robert Lightfoot, Jr.	NASA Associate Administrator	4/15/14
Dr. Michael Freilich	Director, Science Mission Directorate, Earth Science Division	1/24/14
Steve Isakowitz	President, Virgin Galactic	10/23/13
James Wood	KSC Launch Services Program Chief Engineer	9/12/13

# Socialization (cont.)



Name	Position	Date
Dr. Mike Ryschkewitsch	NASA Chief Engineer (at the time)	5/21/13
Dr. Walker, Dr. Tousley, Pam Melroy	DARPA Deputy Director and DARPA Tactical Technology Office	4/29/13
Dr. Antonio Elias	Executive Vice President and Chief Technical Officer, Orbital Sciences Corp.	2/26/13
Doug Shane	President, Scaled Composites (at the time)	1/4/12

### Endorsements





April 11, 2014

Gerald D. Budd, Project Manager Advanced Planning and Partnerships Office, Code Z Neil A. Armstrong Flight Research Center P.O. Box 273, M/S-2701 Edwards, CA 93523-0273

Re: Towed Glider Air-Launch Concept Project

Dear Mr. Budd,

Space Florida is please to express a genuine interest in facilitating the development of the Towed Glider dedicated cubesat launcher concept. We have long believed the small sat market to be as exciting a business opportunity as anything associated with technology. That future is exciting and our job is to assure that future happens here in Florida.

A partnership between Armstrong Flight Research Center and the Kennedy Space Center (KSC), as a pay of the Game Changing Division of NASA's Space Technology Mission Directorate, this technology's development is encouraging. That it would then be handed off to industry to operate is what makes it so intriguing. It is the business opportunity for Florida which so aligns with our charter and our mission.

Space Florida is prepared to sponsor the development of an independent business case for the Towed Glider as a commercially-operated, dedicated cubesat launcher based at Shuttle Landing Facility (SLF). This modest investment will then be used to attract interested industry partners. Additionally, with the existing workforce, resources and facilities already maturing here on the Space Coast to support cubsat development, processing, integration, as well as the potential to process the downlinked data, this concept clearly supports the future we are trying to create here in Florida.

Also, Space Florida can readily facilitate 1) the development of a NASA-funded lease for required shop and office space at SLF for the developmental project; 2) participate in Range coordination for the developmental project; and 3) to participate in securing an industry partner.

Finally, the potential pace of work at the SLF, the further evolution of the cubsat processing and launch capability, as well as the opportunities to engage local university students in this new industry, represent exactly what it is Space Florida was created to support.

We look forward to working with you in the future on this innovative concept.

Sincerely

Frank A. DiBello President & CEO Space Florida

SPACE FLORIDA

505 Odyssey Way • Suite 300 • Exploration Park • FL 32953 www.SpaceFlorida.gov • f: 321.730.5307 • p: 321.730.5301 Space Florida is prepared to sponsor the development of an independent business case for the Towed Glider as a commercially-operated, dedicated cubesat launcher based at Shuttle Landing Facility (SLF). This modest investment will then be used to attract interested industry partners.

Also, Space Florida can readily facilitate 1) the development of a NASA-funded lease for required shop and office space at SLF for the developmental project; 2) participate in Range coordination for the developmental project; and 3) to participate in securing an industry partner.

### **Endorsements**





April 22, 2014

Gerald D. Budd, Project Manager Advanced Planning and Partnerships Office, Code 2 Neil A. Armstrong Flight Research Center P.O. Box 273, M/S-2701 Edwards, CA 93523-0273

Re: Towed Glider Air-Launch Concept Project

Dear Mr. Budd,

Mojave Air and Space Port (MA&SP) is pleased and excited at the opportunity to facilitate the development of the Towed Glider dedicated cubesat launcher concept. We have direct access to special use airspace, a supersonic corridor and a letter on file from FAA authorizing access to space.

MA&SP has engaged in towed aircraft activity for nearly 70 years. Multiple commercial space companies now call Mojave home and more rocket motors have been tested here in Mojave this century than any other location on earth.

MA&SP would be very pleased to further this initiative in any way possible. At least three destinations or original points will be required in North America to reach desired space inclinations. Mojave is a natural for one of these three.

We look forward to working with you in the future on this innovative concept.

Sincerely,

MOJAVE AIR AND SPACE PORT

Stuart O. Witt Chief Executive Officer

SOW/dar

Mojave Air and Space Port (MA&SP) is pleased and excited at the opportunity to facilitate the development of the Towed Glider dedicated cubesat launcher concept.

MA&SP would be very pleased to further this initiative in any way possible.

East Kern Airport District • 1434 Flightline • Mojave, CA 93501 • 661. 824.2433

# **AFRC Historic Capabilities**



- AFRC Tow Experience
  - F-106 tow from C-141A
  - M2-F1 Lifting Body
  - 1/3 Scale Towed Glider
- AFRC Remote Pilot Experience
  - F-15 RPRV
  - HiMAT
  - IKHANA
  - DROID









- AFRC Air Launch Experience
  - Pegasus
  - X-43
  - X-15
  - HL-10
  - M2-F2



**TGALS** is an Integration of Proven Capabilities

# TGALS CIF Activity Photos







**DROID** towing Twin Fuselage Glider









Team photo with Twin Fuselage Glider and MiniSprite Launch Vehicle during fit-check activities

# Flexible Partnering Options



### Military Partner

- Build glider system to augment existing launch capabilities
- Military sizes system to meet their tactical needs

### **Big Commercial Partner**

- Build glider system to be first to market with dedicated small launch capability
- Size glider to cater to projected nearterm small launch market



### **Hybrid Solutions**

 Any 2 or more combine to pay for development of all major parts (glider, tow plane, LVs)







### **NASA Demonstration**

- Tech Demo of minimum cost proof of concept system (likely sub-scale)
  - Tech Transfer to industry to commercialize as they see fit

### **Multiple Small Partners**

- Each contributes part of the glider system to open small launch market
- Enjoy first to market status without huge capital outlays

# Excess Thrust Required by Tow Airplane as a function of Carry Efficiency and Tow Glider Max L/D



